

# Curved Beams

## Derivation of Stress Equation

Mechanics of Solids-2

Lecture # 5

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# Summary of Previous Lectures

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- Thin and Thick Walled Cylinders
- Applications of Thin and Thick Walled Cylinders
- Three Different forms of Stresses in Cylinders
  - Longitudinal Stresses
  - Longitudinal Strains
  - Change in Length
  - Hoop or Circumferential Stresses
  - Circumferential Strain
  - Change in Diameter
  - Change in Volume
  - Radial Stresses
- Derivation of Formulae for Stresses in Thin and Thick Walled Cylinders
- Application of Longitudinal, Circumferential, and Radial Stresses on Real World Engineering Problems.

# Table of Content for Lecture # 5

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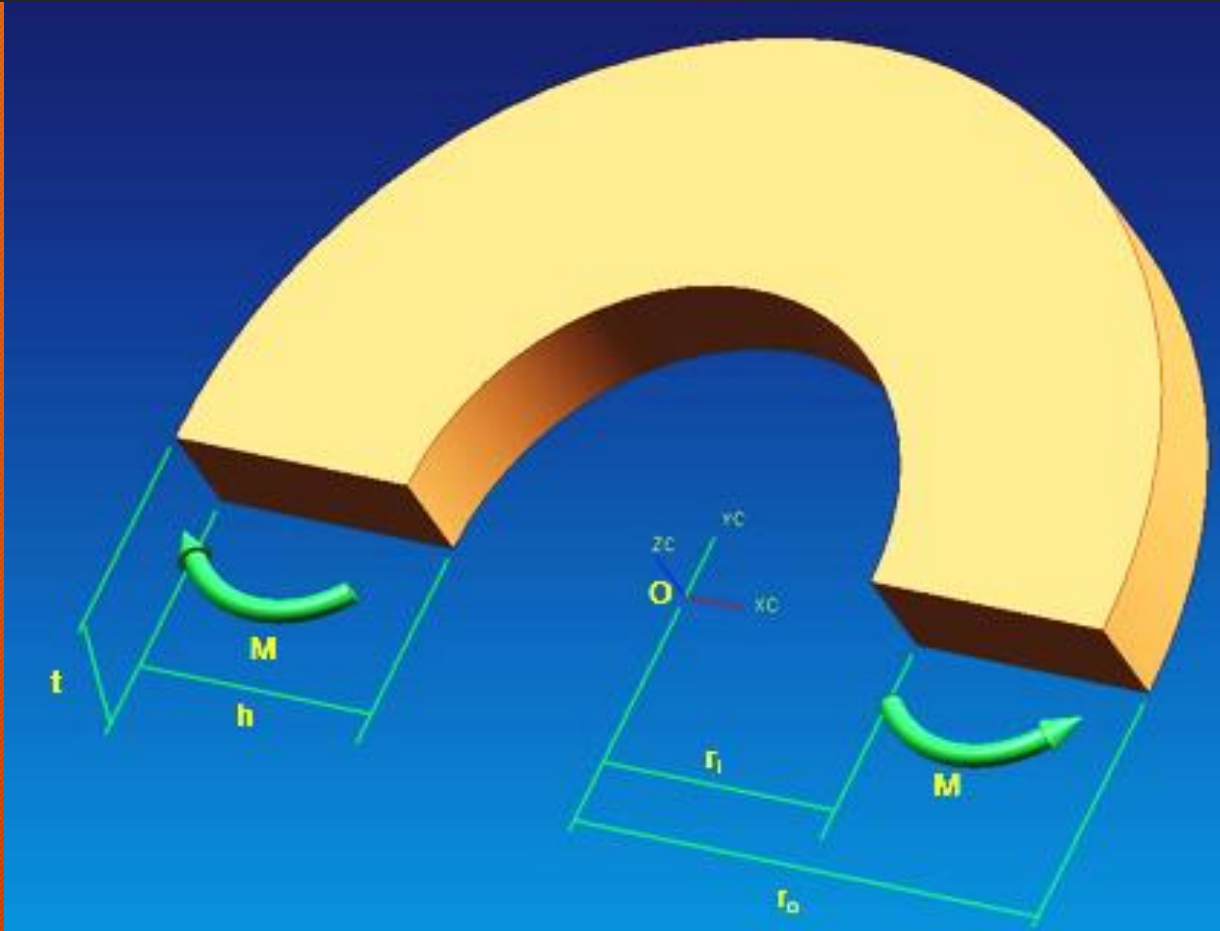
- Derivation of Stresses in Curved Beams.



# Curved Beams

## Derivation of Stress

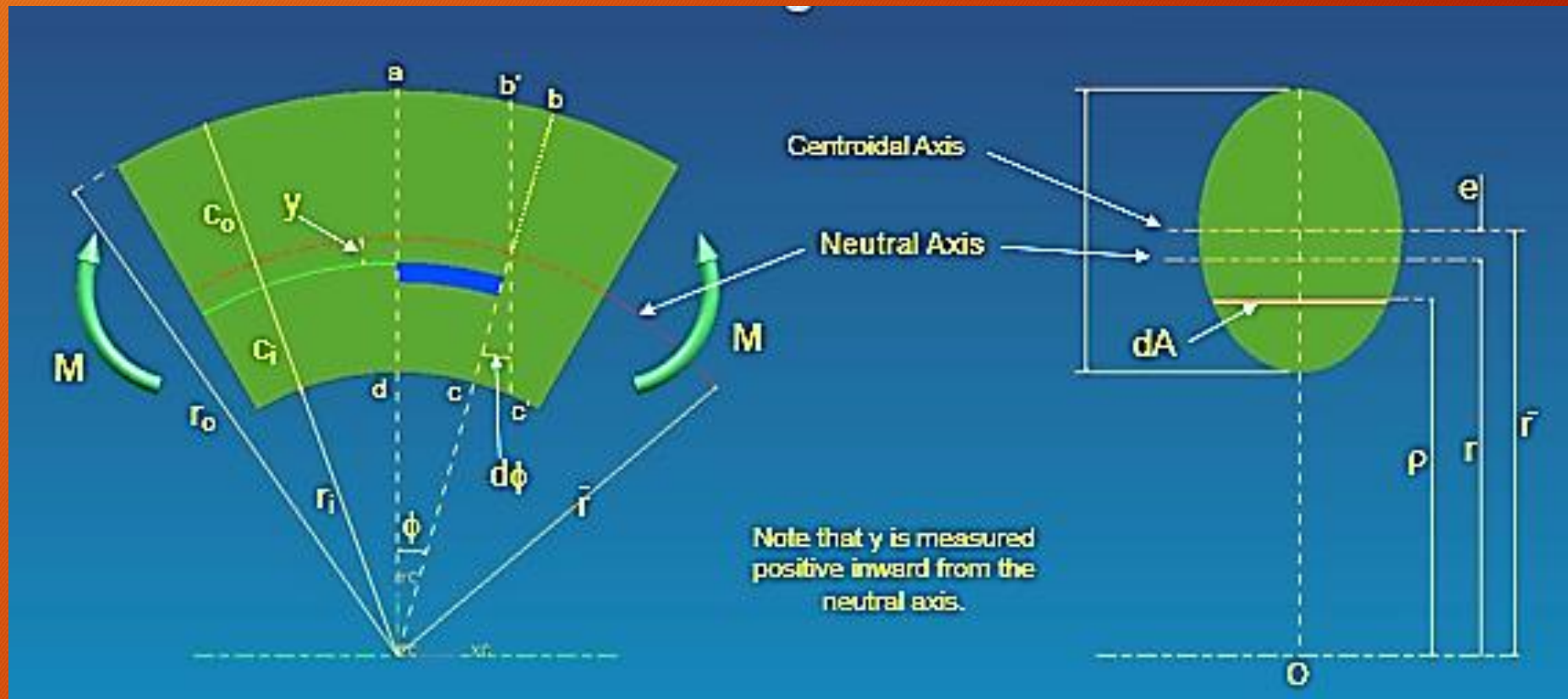
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# Curved Beams

## Derivation of Stress

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# Curved Beams - Derivation of Stress

## Curved Members in Flexure

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The distribution of stress in a curved flexural member is determined by using the following assumptions.

1. The **material** of the beam is perfectly **homogenous** i.e. same material throughout, and **isotropic** i.e. equal elastic properties in all the directions.
2. The **cross section** has an axis of symmetry in a plane along the length of the beam.
3. The **material** of the beam **obeys hooks' law**.
4. Plane **cross sections** remain **plane after bending**.
5. The **modulus of elasticity** is the **same** in tension as in compression.

It will be found that the **neutral axis** and the **centroidal axis** of a curved beam, unlike a straight beam, are not coincident and also that the stress does not vary linearly from the neutral axis. The notation shown in the above figures is defined as follows:



# Curved Beams - Derivation of Stress

## Curved Members in Flexure

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$r_o$	=	radius of outer fiber
$r_i$	=	radius of inner fiber
$h$	=	depth of section
$c_o$	=	distance from neutral axis to outer fiber
$c_i$	=	distance from neutral axis to inner fiber
$r$	=	radius of <b>neutral</b> axis
$\bar{r}$	=	radius of <b>centroidal</b> axis
$e$	=	distance from centroidal axis to neutral axis

To begin, we define the element  $abcd$  by the angle  $\phi$ . A bending moment  $M$  causes section  $bc$  to rotate through  $d\phi$  to  $b'c'$ . The strain on any fiber at distance  $\rho$  from the center  $O$  is

$$\epsilon = \frac{\delta l}{l} = \frac{(r - \rho) d\phi}{\rho \phi}$$

# Curved Beams - Derivation of Stress

## Curved Members in Flexure

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The normal stress corresponding to this strain is

$$\sigma = \varepsilon E = \frac{E(r - \rho)d\phi}{\rho\phi} \quad (1)$$

Since there are no axial external forces acting on the beam, the sum of the normal forces acting on the section must be zero. Therefore

$$\int \sigma dA = E \frac{d\phi}{\phi} \int \frac{(r - \rho)dA}{\rho} = 0 \quad (2)$$

Now arrange Eq. (2) in the form

$$E \frac{d\phi}{\phi} \left( r \int \frac{dA}{\rho} - \int dA \right) = 0 \quad (3)$$



# Curved Beams - Derivation of Stress

## Curved Members in Flexure

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$$E \frac{d\phi}{\phi} \left( r \int \frac{dA}{\rho} - \int dA \right) = 0 \quad (3)$$

and solve the expression in parentheses. This gives

$$r \int \frac{dA}{\rho} - A = 0 \quad \text{or} \quad r = \frac{A}{\int \frac{dA}{\rho}} \quad (4)$$

This important equation is used to find the location of the neutral axis with respect to the center of curvature  $O$  of the cross section. The equation indicates that the neutral and the centroidal axes are not coincident.

# Curved Beams - Derivation of Stress

## Curved Members in Flexure

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Our next problem is to determine the stress distribution. We do this by balancing the external applied moment against the internal resisting moment. Thus, from Eq. (2),

$$\int (r - \rho) \sigma dA = E \frac{d\phi}{\phi} \int \frac{(r - \rho)^2 dA}{\rho} = M \quad (5)$$

Since  $(r - \rho)^2 = r^2 - 2\rho r + \rho^2$ , Eq. (5) can be written in the form

$$M = E \frac{d\phi}{\phi} \left( r^2 \int \frac{dA}{\rho} - r \int dA + \int \rho dA \right) \quad (6)$$

Note that  $r$  is a constant; then compare the first two terms in parentheses with Eq. (4). These terms vanish, and we have left

$$M = E \frac{d\phi}{\phi} \left( -r \int dA + \int \rho dA \right)$$

The first integral in this expression is the area  $A$ , and the second is the product  $rA$ . Therefore

$$M = E \frac{d\phi}{\phi} (r - r)A = E \frac{d\phi}{\phi} eA$$

Now, using Eq. (1) once more, and rearranging, we finally obtain  $\sigma = \frac{My}{Ae(r - y)}$

# Curved Beams - Derivation of Stress

## Curved Members in Flexure

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This equation shows that the stress distribution is hyperbolic. The algebraic *maximum* stresses occur at the inner and outer fibers and are

$$\sigma_i = \frac{Mc_i}{Aer_i} \quad \sigma_o = \frac{Mc_o}{Aer_o} \quad (7)$$

The sign convention used is that  $M$  is positive if it acts to straighten on the beam. The distance  $y$  is positive inwards to the center of curvature and is measured from the neutral axis. It follows that  $c_i$  is positive and  $c_o$  is negative.

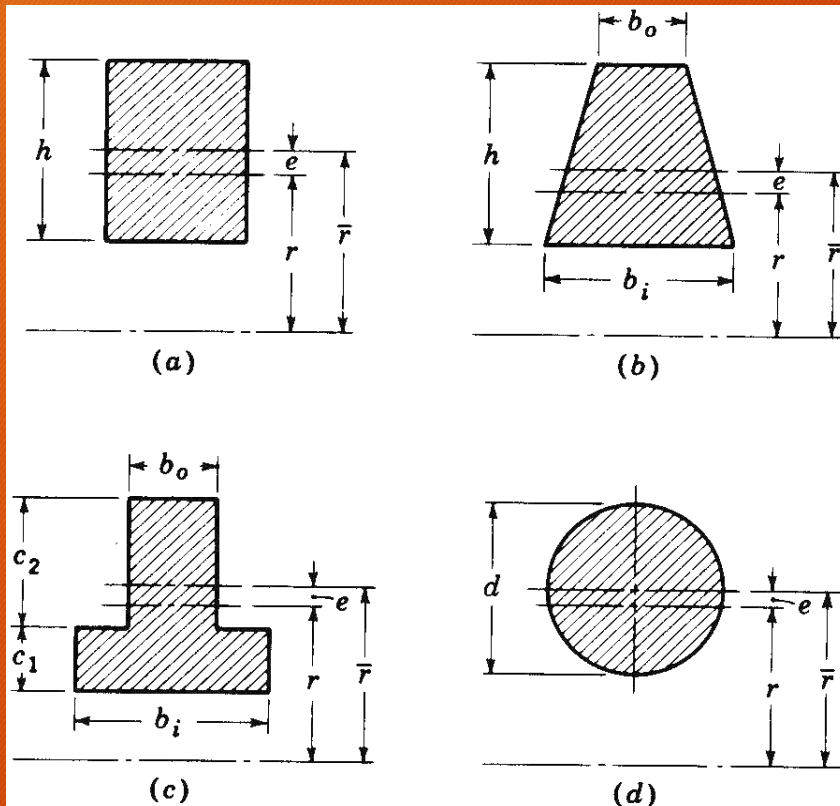
These equations are valid for pure bending. In the usual and more general case such as a crane hook, the U frame of a press, or the frame of a clamp, the bending moment is due to forces acting to one side of the cross section under consideration. In this case the bending moment is computed about the *centroidal axis*, *not* the neutral axis. Also, an additional axial tensile ( $P/A$ ) or compressive ( $-P/A$ ) stress must be added to the bending stress given by Eq. (7) to obtain the resultant stress acting on the section.



# Curved Beams - Derivation of Stress Formulas for Some Common Sections

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Sections most frequently encountered in the stress analysis of curved beams are shown below.



For the rectangular section shown in (a), the formulae are

$$\bar{r} = r_i + \frac{h}{2} \quad \text{and} \quad r = \frac{h}{\ln(r_o/r_i)}$$

For the trapezoidal section in (b), the formulae are

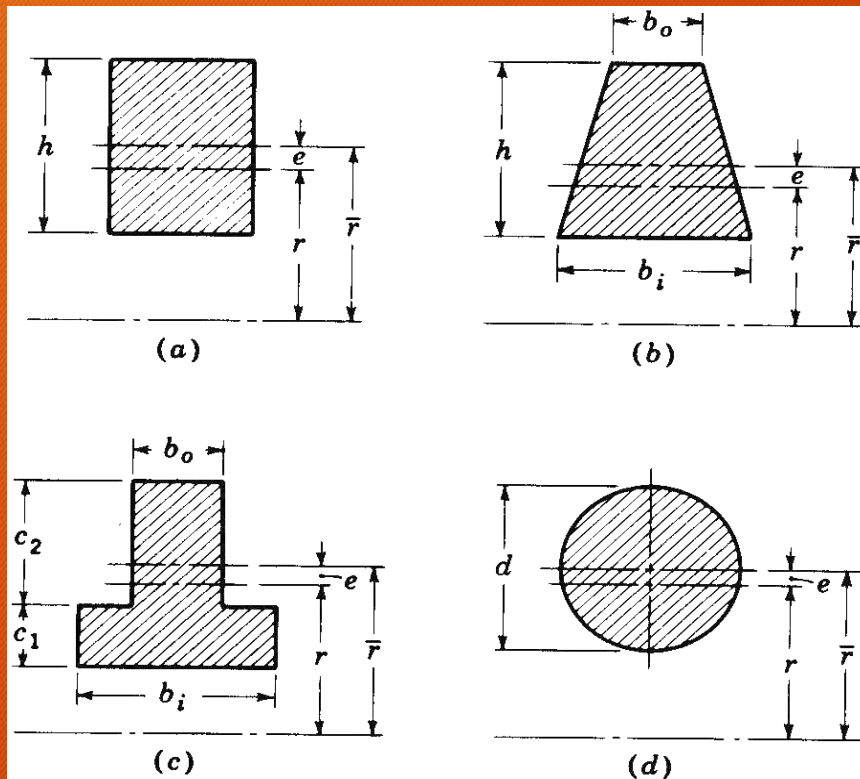
$$\bar{r} = r_i + \frac{h}{3} \frac{b_i + 2b_o}{b_i + b_o}$$

$$r = \frac{A}{b_o - b_i + \left[ \frac{(b_i r_o - b_o r_i)}{h} \right] \ln(r_o/r_i)}$$

# Curved Beams - Derivation of Stress Formulas for Some Common Sections

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Sections most frequently encountered in the stress analysis of curved beams are shown below.



For the T section in we have

$$r = r_i + \frac{b_i c_1^2 + 2b_o c_1 c_2 + b_o c_2^2}{2(b_o c_2 + b_i c_1)}$$

$$r = \frac{b_i c_1 + b_o c_2}{b_i \ln[(r_i + c_1)/r_i] + b_o \ln[r_o/(r_i + c_1)]}$$

The equations for the solid round section of Fig. (d) are

$$r = r_i + \frac{d}{2}$$

$$r = \frac{d^2}{4(2\bar{r} - \sqrt{4\bar{r}^2 - d^2})}$$

# Thank-you for Listening!

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**Be yourself. (Shad)**